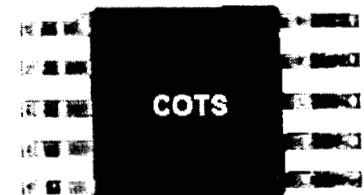


Military & Aerospace Applications of Programmable Devices and Technologies Conference

**Goddard Space Flight Center
1998**

JPL's Commercial Off-The-Shelf (COTS) Program

**A Step Towards Infusion of Reliable COTS Plastic
Parts in NASA Flight Hardware**



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The Meaning of COTS

- “Buy and Fly”
- “Procuring via catalog part number to QML-V standards”
- “Procurement is performed without formal specification”
- “The usage of any COTS equipment does not constitute any waiver to fundamental applicable requirements”

JPL Interpretation:

COTS are parts whose specification is manufacturer-controlled as opposed to traditional “Hi-Rel” parts whose specification was Government or customer-controlled



Why Put COTS Plastic Parts in Space ?

- 1. The availability of COTS plastic parts is proliferating.**
- 2. COTS plastic parts performance capabilities continue to increase (e.g. processing power & high density memories)**
- 3. COTS plastic parts enable reduction of hardware weight and volume**
- 4. COTS plastic parts typically cost much less than ceramic**
- 5. COTS plastic parts have been reported to demonstrate good to excellent reliability in commercial and aerospace applications**
- 6. Often they are the only option available to using state-of-the-art technologies**



JPL's Concerns About Using COTS Plastic Parts

- **Reliability/RH of Plastic in Space Applications**
- **Non Rad Hard Designed (maybe Rad Tolerant)**
- **Narrow Temperature Range for Commercial Grade**
- **Process/Designs Change Frequently**
- **Lack of Lot Traceability & Uniformity**

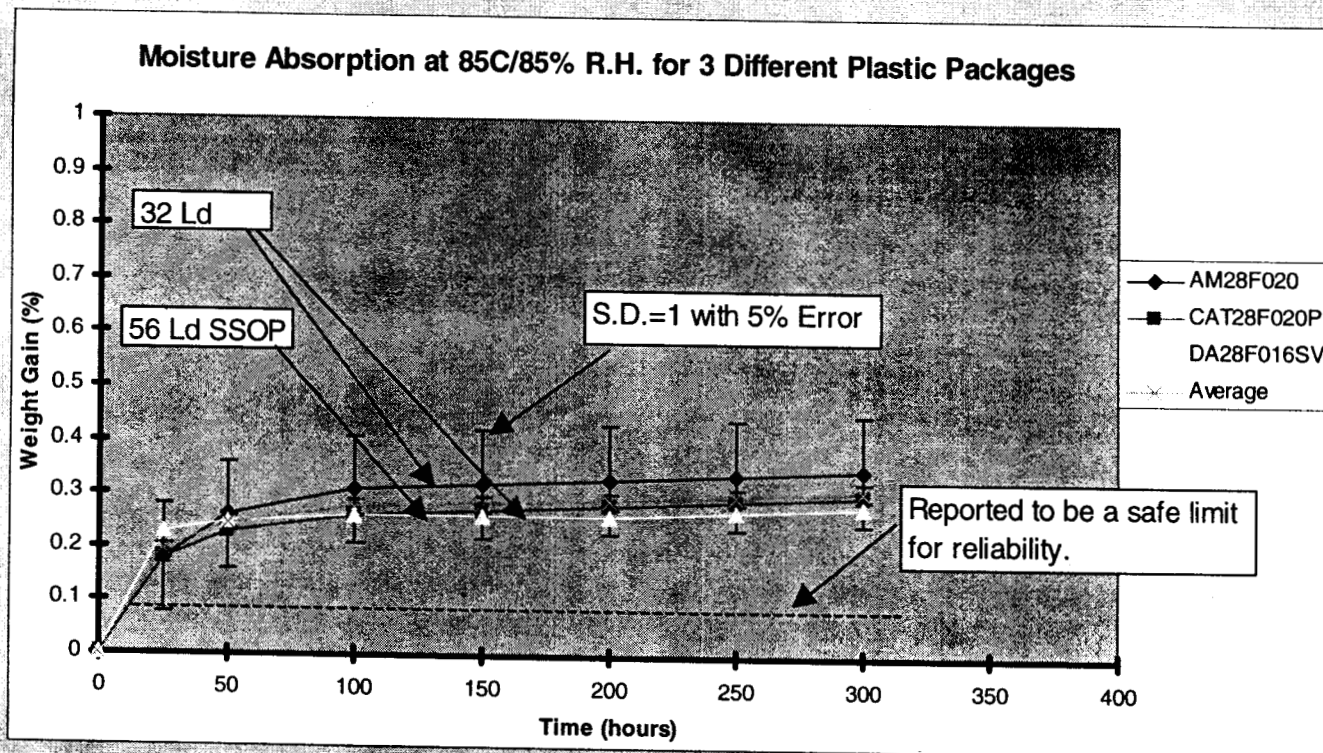
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Assessment Options for COTS Plastic Parts & Their Relative Test

• Temperature/ Humidity	➡	Corrosion	(\$)
• Temperature Cycling	➡	Assembly Defects	(\$\$)
• Moisture Absorption	➡	Popcorning	(\$\$)
• Radiation	➡	TID Degradation	(\$\$\$\$)
• Outgassing	➡	Condensables	(\$)
• Glass Transition	➡	Epoxy Stability	(\$\$)
• Delamination	➡	Voids/Stresses	(\$)
• Upscreening/Burn-in	➡	Performance/Reliability	(\$\$\$\$)
• DPA	➡	Manufacturing Quality	(\$\$)

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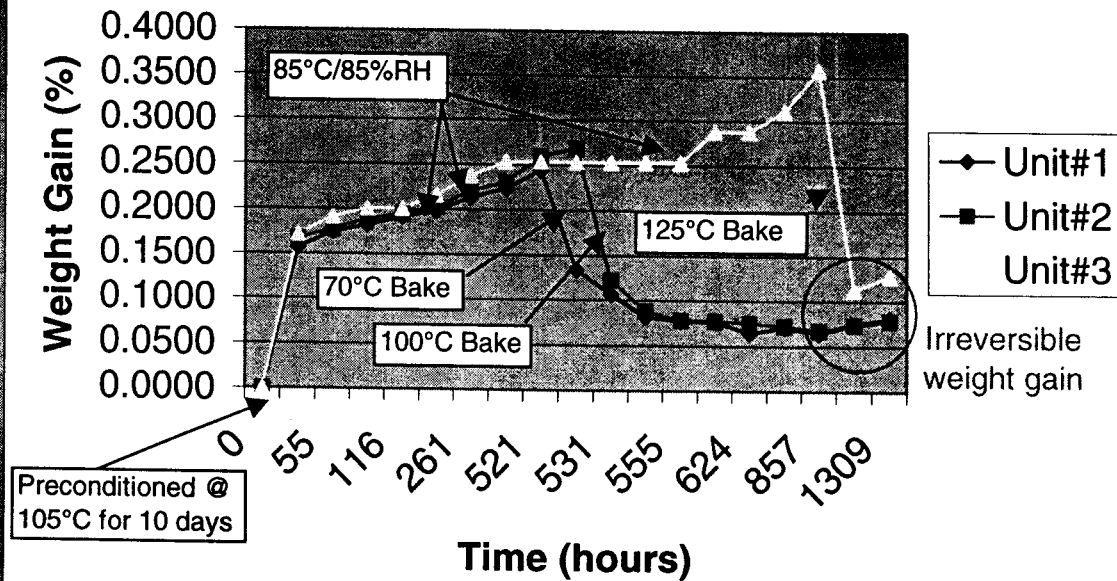


Conclusion: Most if not all plastic parts will absorb moisture >> 0.1% weight gain.

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**Moisture Absorption/Desorption for Intel
56 Ld SSOP Package**



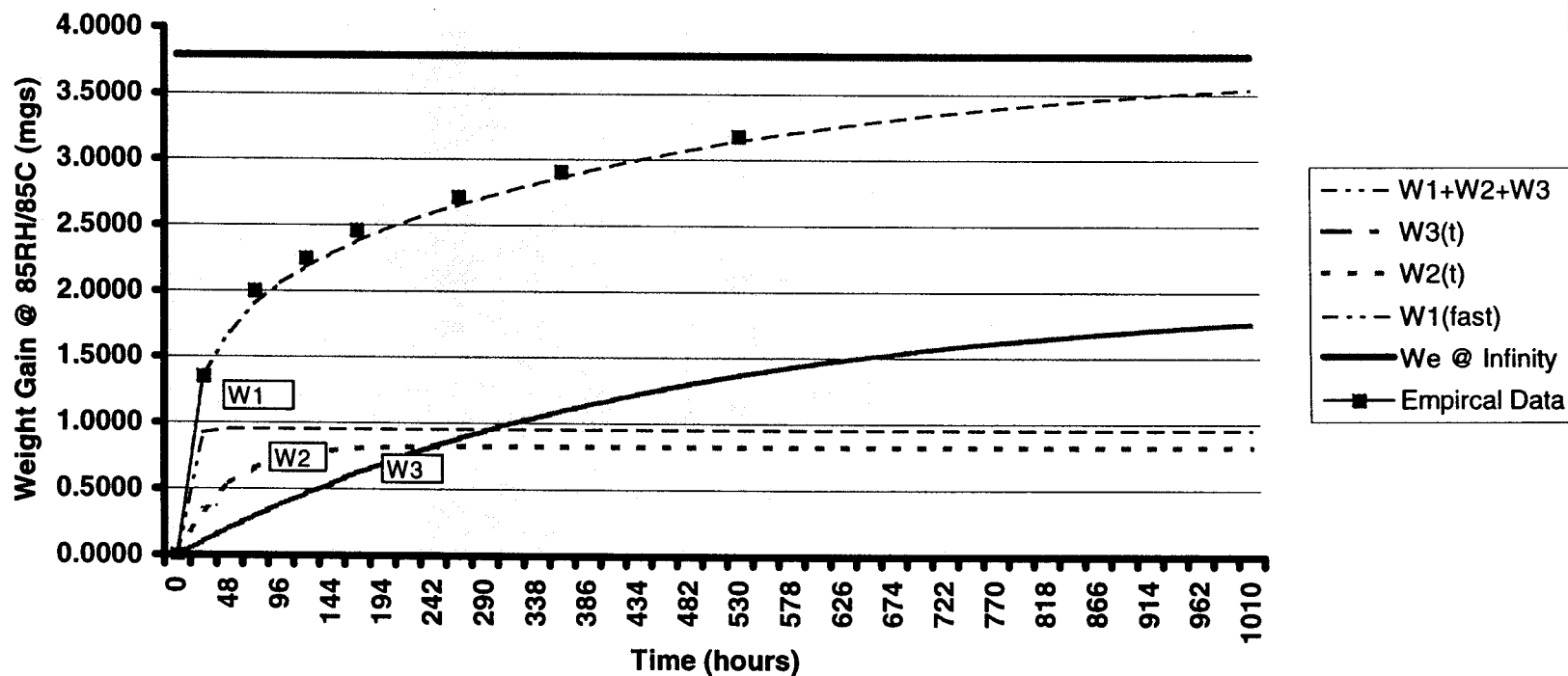
Note: Weight Gain (%) = $(W_f - W_i) / (W_i - W_d) \times 100$
Weight Loss (%) = $(W_d - W_i) / (W_f - W_i) \times 100$

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Moisture Multiple Absorption Model for SCR265 (Plastic)

$$W(t) = W_e(1 - e^{-kt})$$



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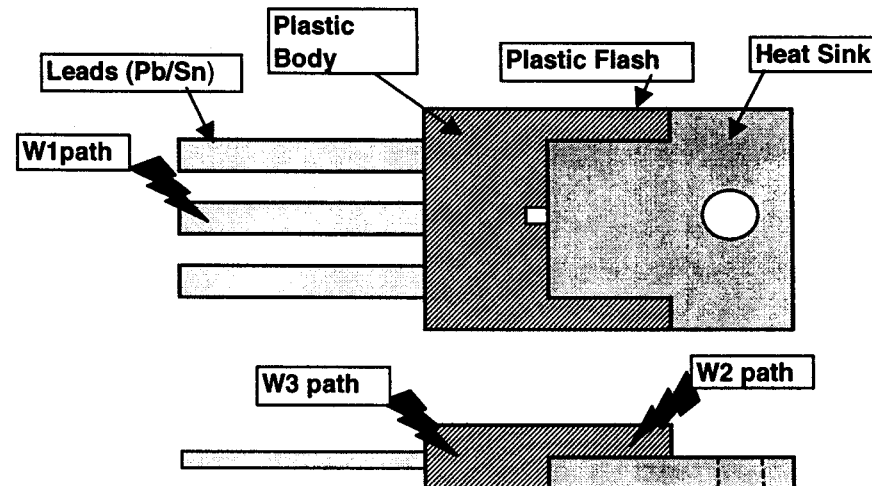


85%RH/85°C Moisture Absorption Mechanisms for SCR265

➤ **W1(t): Fast Irreversible Weight Gain ≤ 24 hours (~1mg)**

➤ **W2(t): Intermediate Reversible Weight Gain, 80 to 140 hours**

➤ **W3(t): Slowest Reversible Weight Gain, Reaches W_e @ $t=\infty$**
SCR265 Package



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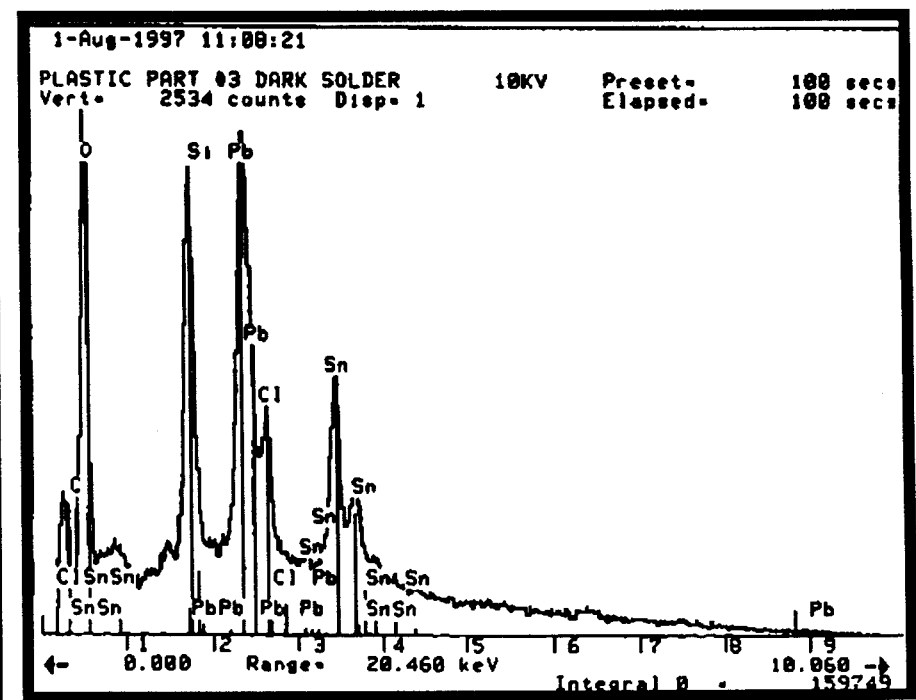
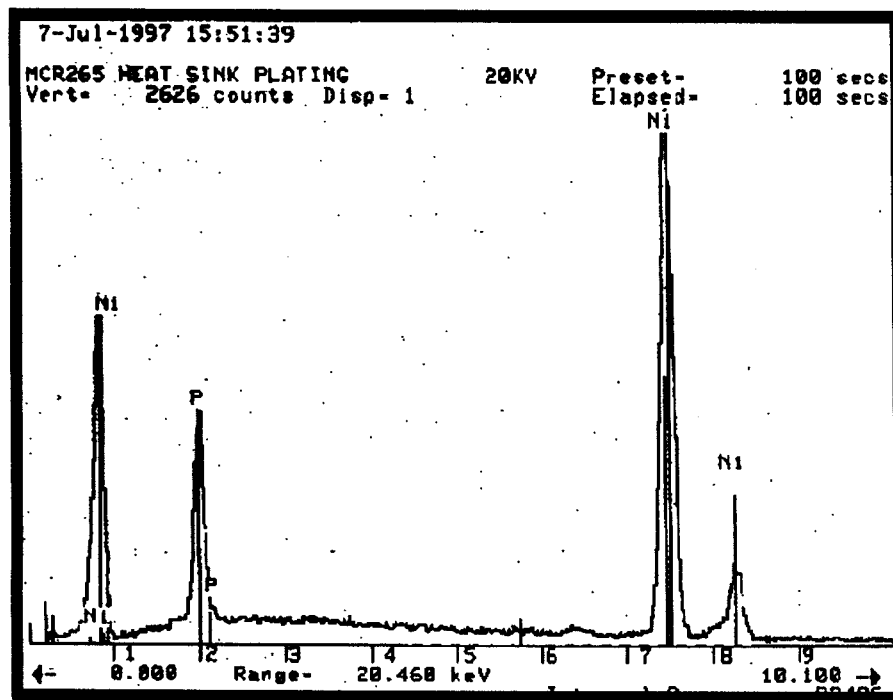
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**Nickel Plated Heatsink
Shows No Oxidation**

Post 85%RH/85°C for SCR265

Leads Show Extreme Oxidation

➔ **W1(t)**

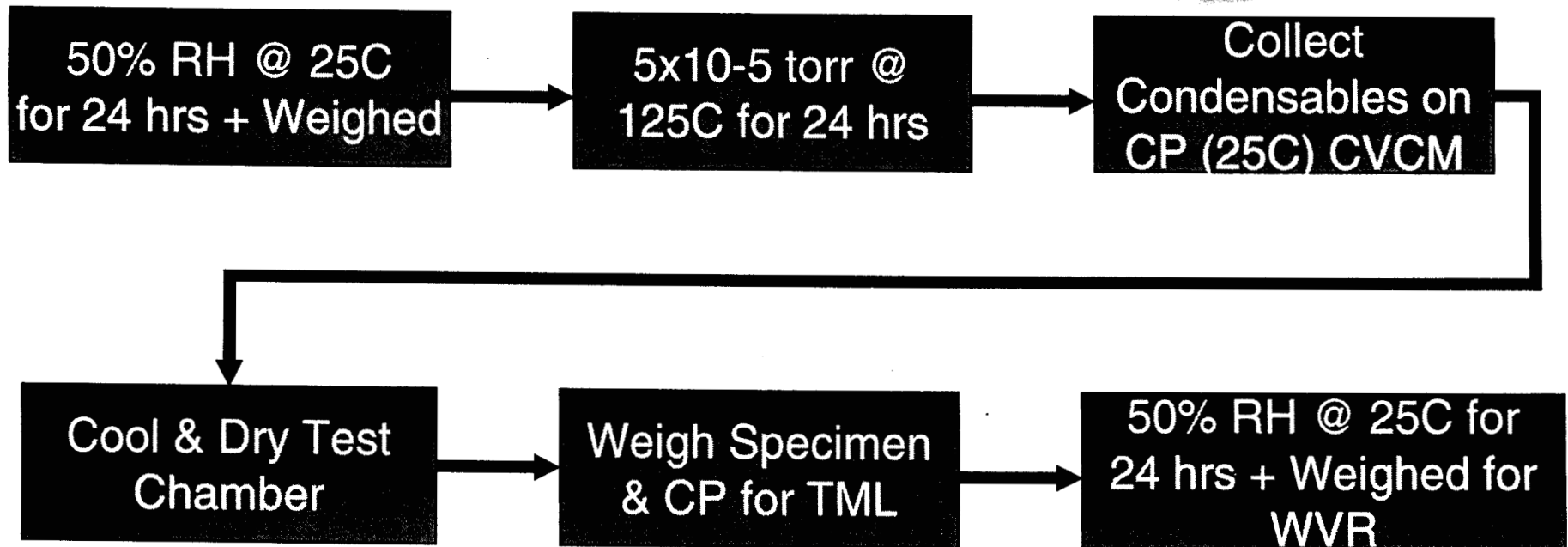


Conclusion: Weight gain is solely attributed to oxidation of leads. The internal chip has miniscule Al area available for oxidation because of Cu intermetallic bonding to the Al.

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Outgassing Test Flow for Plastic Packages



Ref: ASTM E595-93

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Outgassing Results of Plastic Packages

Material	MCR			7612382FBA, E24, DA28F016SV, K8055, U6240332			AM28F020-150PC, 9618FBB			CSI, CAT28F020F, 1-15 09550B		
Part	Motorola SCR			Intel 16 M Flash Memory			AMD 2M Flash Memory			Catalyst 2M Flash Memory		
Sample No.	5	6		7	8	a	9	10		11	24	
WT. Loss %	0.45	0.46	0.45	0.23	0.22	0.22	0.41	0.45	0.43	0.40	0.41	0.40
Water Vapor Recovered, WVR,	0.28	0.25	0.26	0.14	0.11	0.12	0.19	0.17	0.18	0.21	0.18	0.19
%ML (WT, LOSS- WVR) %	0.17	0.21	0.19	0.09	0.11	0.10	0.22	0.28	0.25	0.19	0.23	0.21
CVCM %	0.04	0.08	0.06	0.02	0.01	0.01	0.03	0.05	0.04	0.04	0.04	0.04
DEPOSIT on CP	Opaque			Negligible			Opaque			Opaque		
FTIR Results	Amine cured epoxy			Anhydride cured epoxy			Amine cured epoxy			Amine cured epoxy		

Conclusion: All materials passed. These tests are suited for lot-to-lot comparisons, tracking manufacturing continuity/changes, and measuring absorbed moisture at a known environment.

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507

Radiation Results on Plastic Parts

Moisture Absorption / Bake for Intel DA28F016SV in Plastic Package

(0.6 μ m ETOX IV Process Technology)

Conditions: Test Temperature = 25°C, Vdd = 5.0V, Vpp = 5.0V

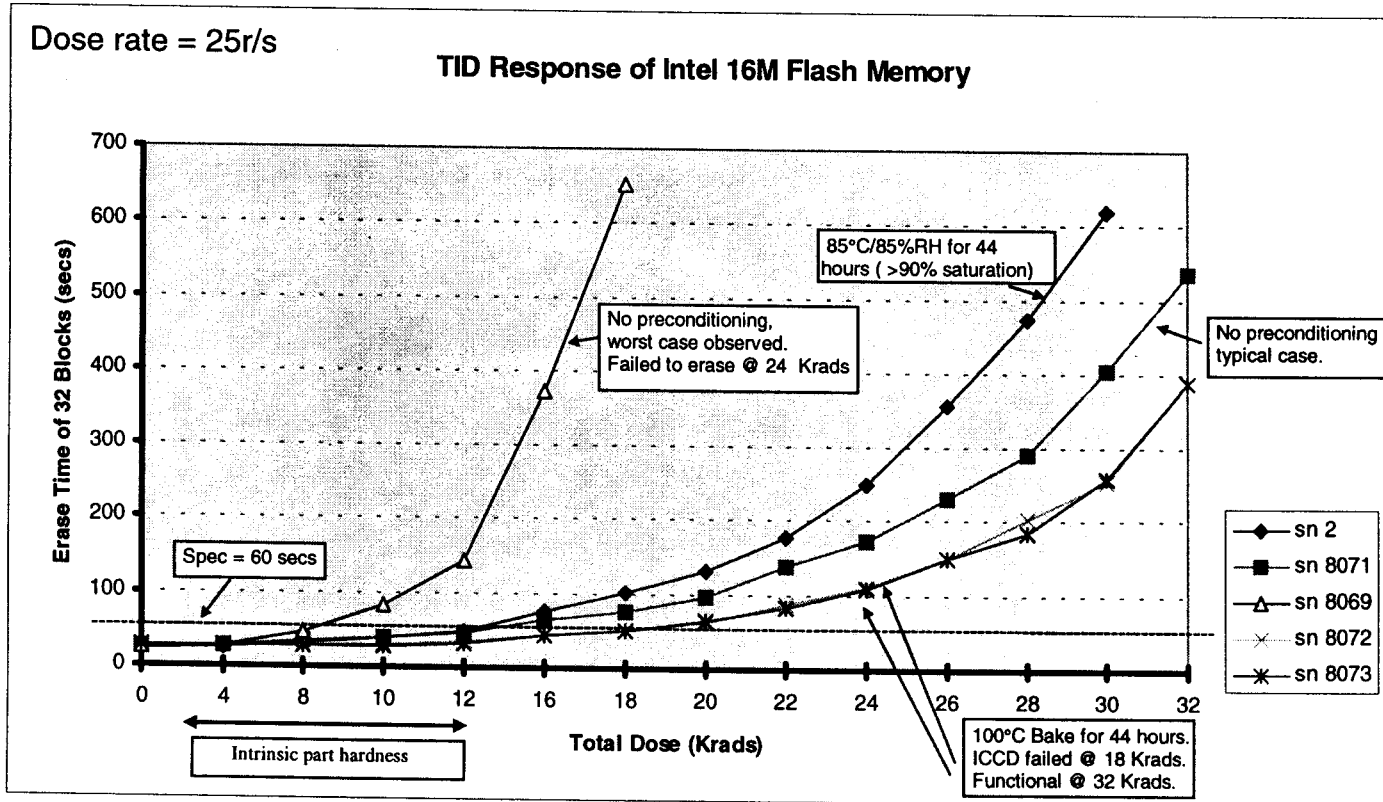


Figure 1
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IDENTIFY & REVIEW REQUIREMENTS

COST & TAILOR OBJECTIVES

DPA

C-SAM

ELECTRICAL

TEMP CYCLE

ELECTRICAL

Burn-in

ELECTRICAL

**Part Level
Screening**

**ASSEMBLE
HARDWARE**

**ASSEMBLY
TEST**

**ASSEMBLY
QUALIFICATION**

**FLIGHT
READY**

COTS Plastic Infusion Baseline Flow



In Summary

- Using COTS plastic parts without understanding their pedigree can lead to mission delay or worst ➡ **Mission Failure**
- A methodology is in place in Office 507 to help JPL users of plastic parts ascertain their risk and acceptance for Space Application
- Work is underway in Office 507 to evaluate **all risk factors** when using COTS plastic parts (quality, reliability, radiation, package).
- JPL is currently infusing plastic parts into flight hardware utilizing a **Better-Faster-Cheaper Program** tailored individually for each Project